

# Hybrid Propulsion System and Aircraft-level Performance Simulation

F. Oliviero, V. Cipolla

Department of Civil and Industrial Engineering, University of Pisa

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Stuttgart, 18-19<sup>th</sup> February 2016



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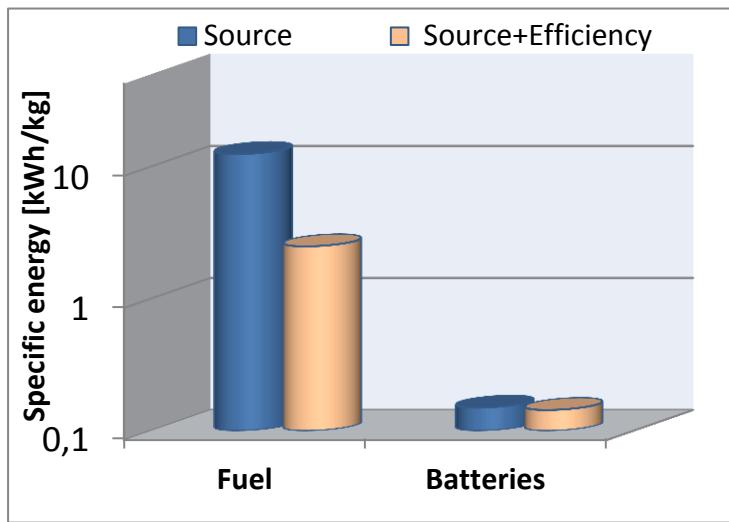
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- Introduction to performance estimation
- Preliminary analysis of flight performance
- HyPSim (Hybrid Plane Simulator)

# Background

Analysis of the flight performance when a non conventional propulsion system is considered.  
Dependence on the energy sources, system architecture, control logics.

Fuel vs battery (today)



## Purposes of the analyses:

- Sizing of the propulsion system components for a given reference design mission (cruise speed, cruise altitude, climb rate, flight programs, etc.);
- Sensitivity analysis of propulsion system performance to different mission parameters;
- Definition of strategies to optimize energy management in each flight segment;
- Definition of critical flight conditions.

# Performance analysis for *Hypstair*

Within the *Hypstair* project, two design stages has been faced:

## Preliminary Performance Analysis Tool

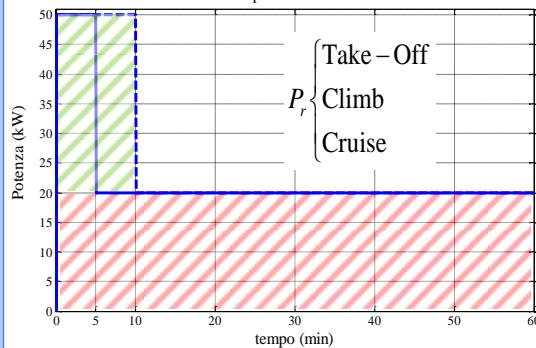
- A set of analytical models to determine the flight performance on the basis of a reference mission profile
- Simple, proper reliability
- Global performance estimation
- Analysis of the battery utilization on specific flight segments (climb, take off) depending on main parameters (aircraft trim)

## The Hybrid Plane Simulator (HyPSim)

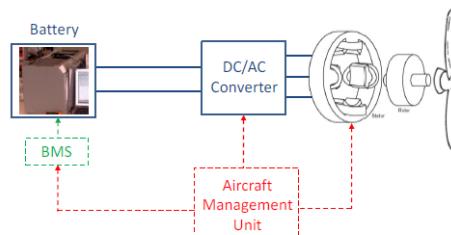
- Introducing the human-in-the-loop effects
- Introducing new mission profiles
- Identification of possible critic conditions in specific flight segments
- Evaluate information on the HMI panel

# Performance analysis for Hypstair

## Performance analysis



## Component sizing

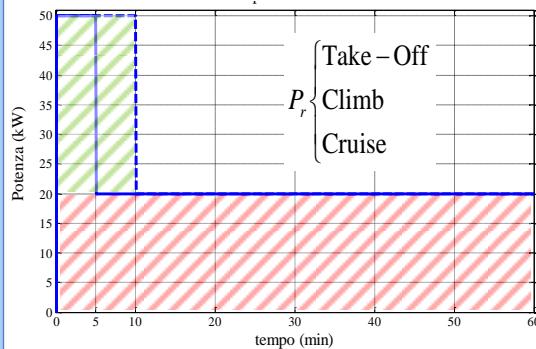


## Testing

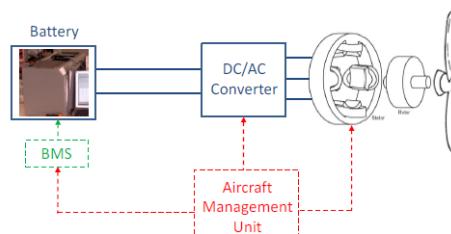


# Performance analysis for Hypstair

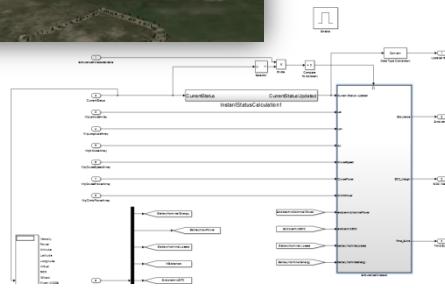
## Performance analysis



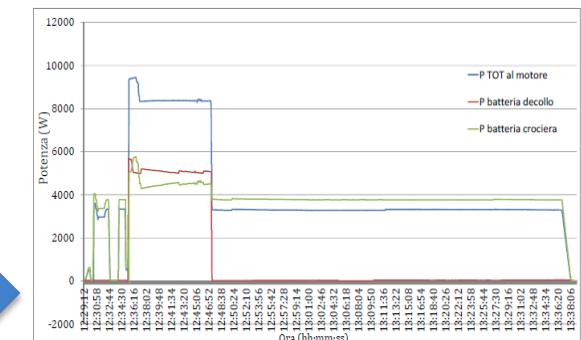
## Component sizing



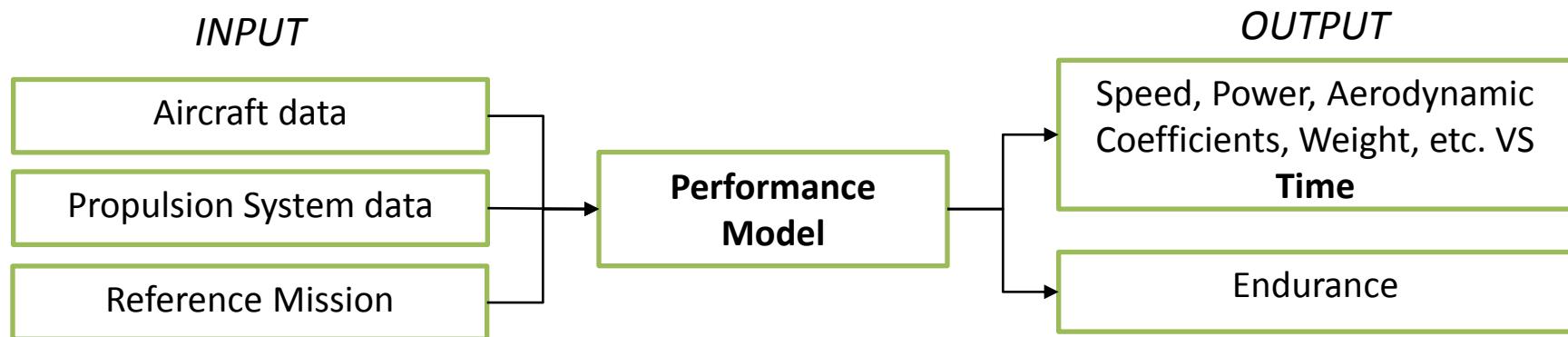
## Simulator



## Testing



# Preliminary Performance Analysis

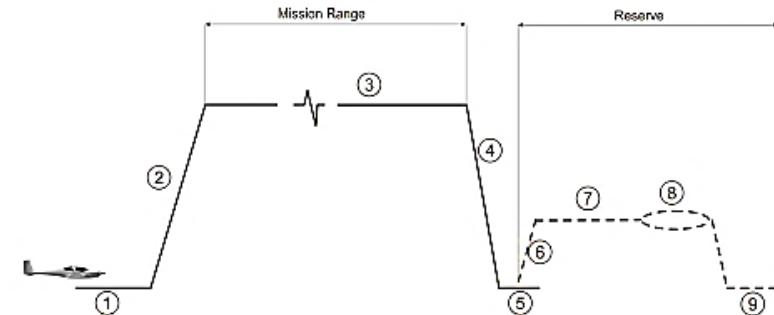


Main features:

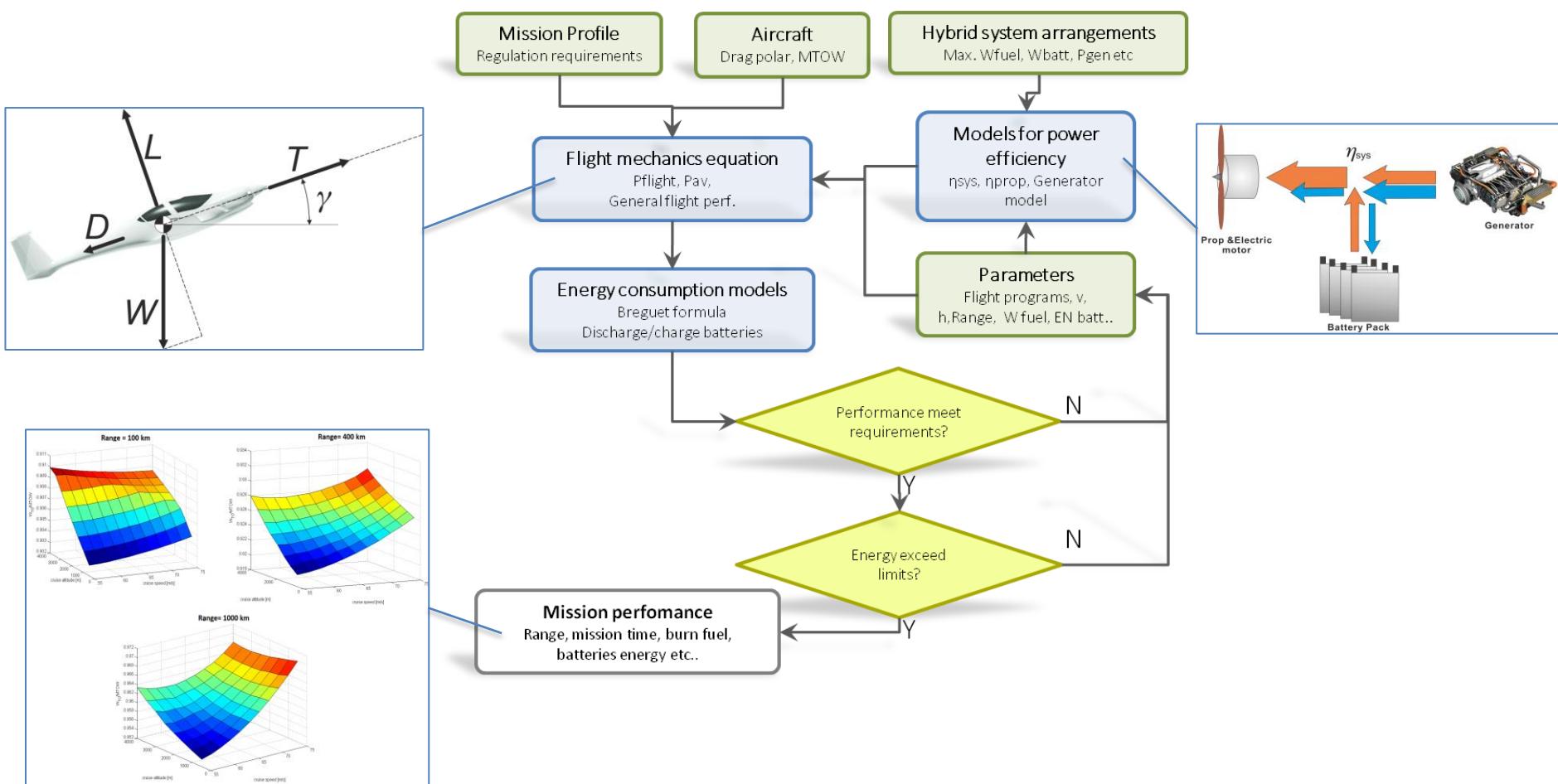
- Need to define the entire mission at input level
- Very simple aerodynamics (polar drag)

Two “optimization” problem to solve:

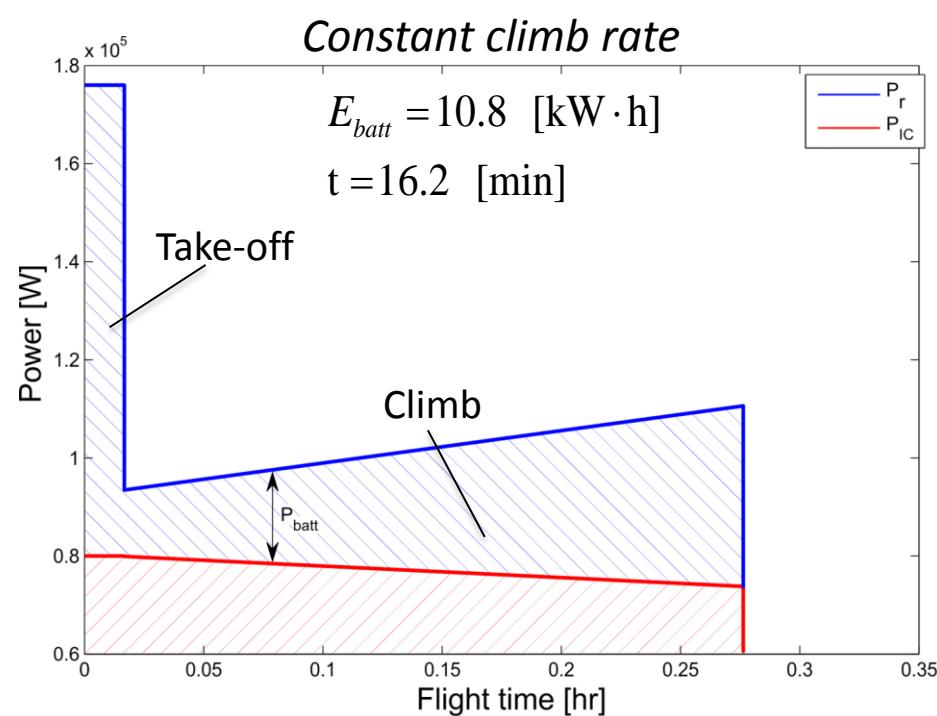
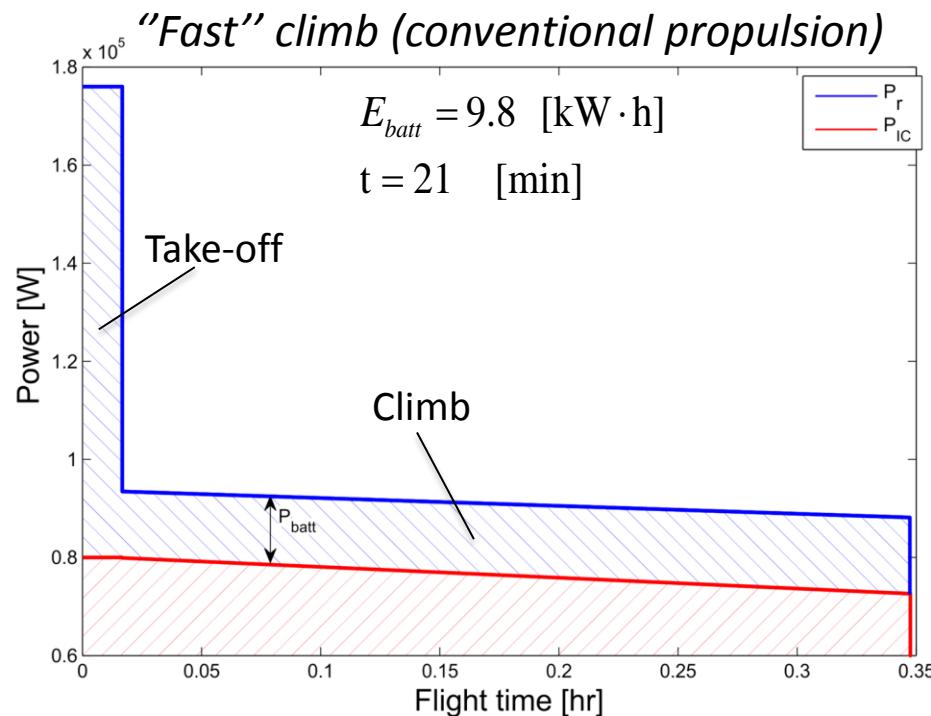
- Evaluation of the maximum flight range achievable with given amount of available energy (fuel + batteries) at take-off;
- Evaluation of the minimum energy amount (fuel + batteries) required at take-off in order to fly for a given range.



# Preliminary Performance Analysis

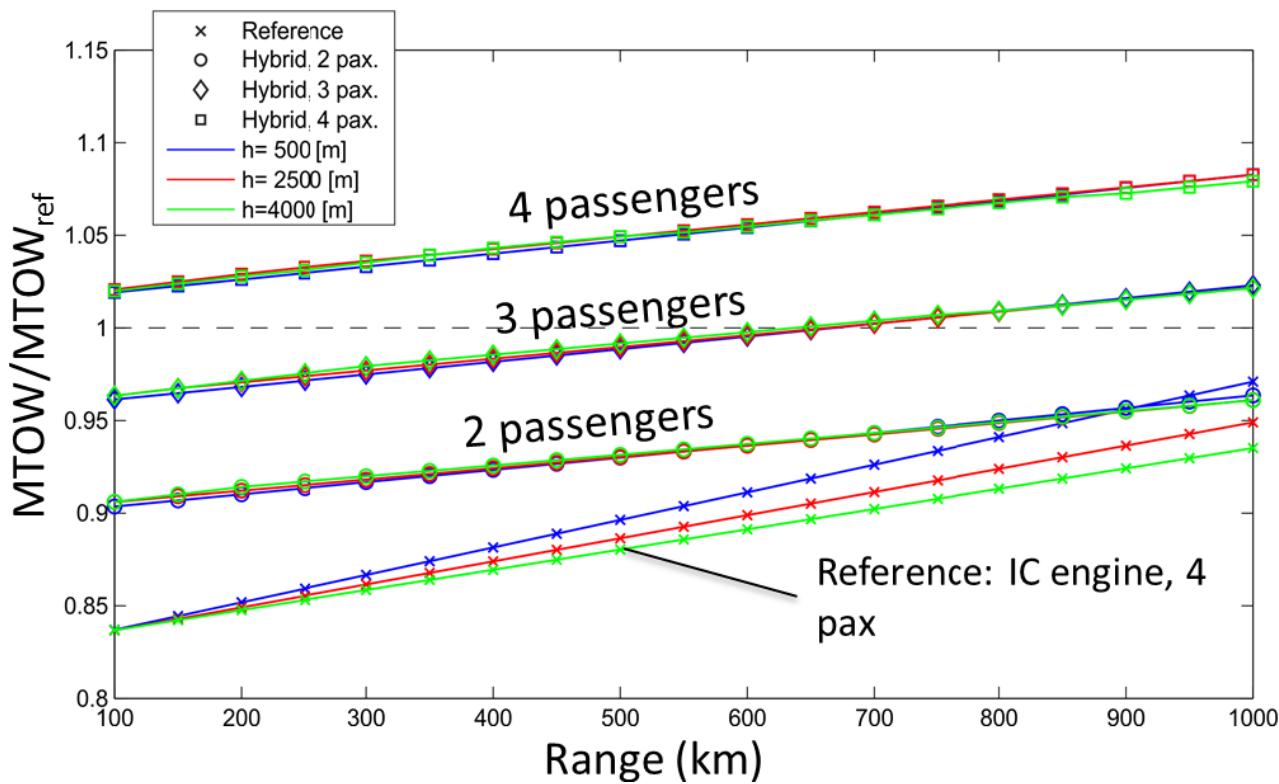


# Effects on specific flight segments



Performance during climb depend strongly on the utilization of the battery: the definition of “optimum” climb remains an open problem

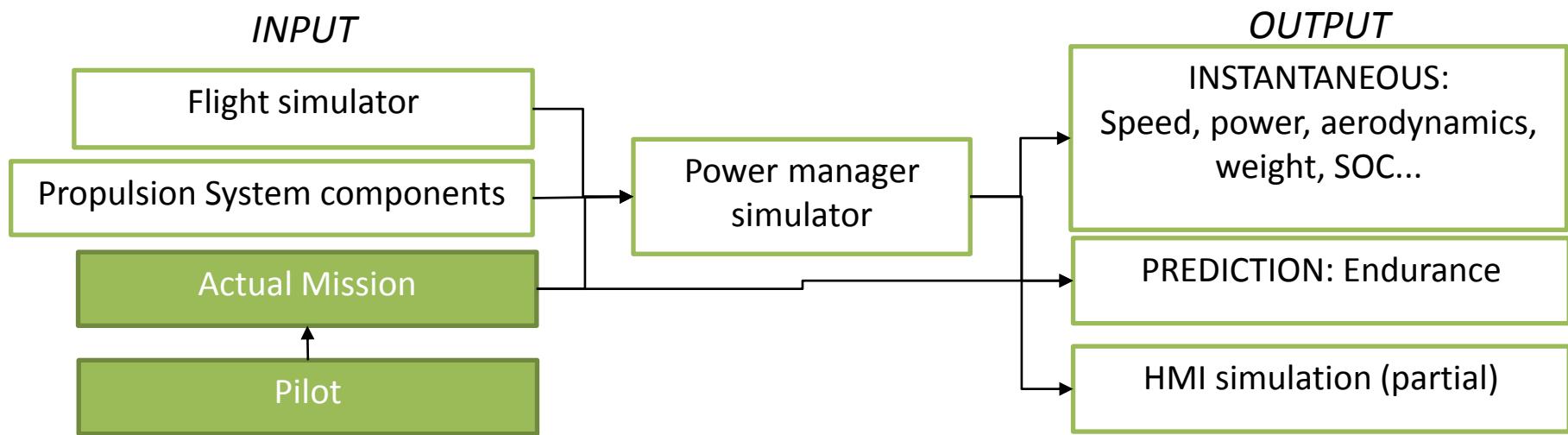
# Effects on global performance



- Batteries increase empty weight, limiting payload capabilities

- Weak MTOW vs Range dependency
- Low sensitivity to cruise altitude
- More flexibility in mission definition

# HyPSim: introduction

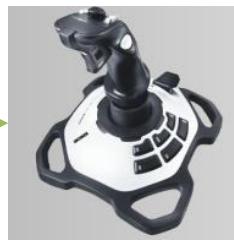


Main purposes:

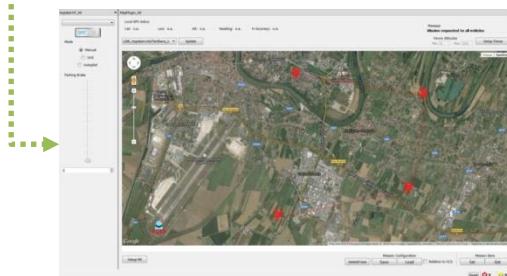
- Introducing the human-in-the-loop effects and different mission profiles;
- Determination of possible critical flight conditions;
- Possibility to include/modify components behavior;
- Guarantee the reproducibility of the experiments.

# HyPSim: general architecture

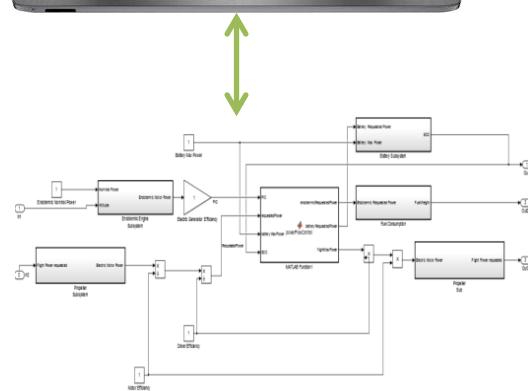
User interaction



Joystick



Mission Planner  
(in-house developed)



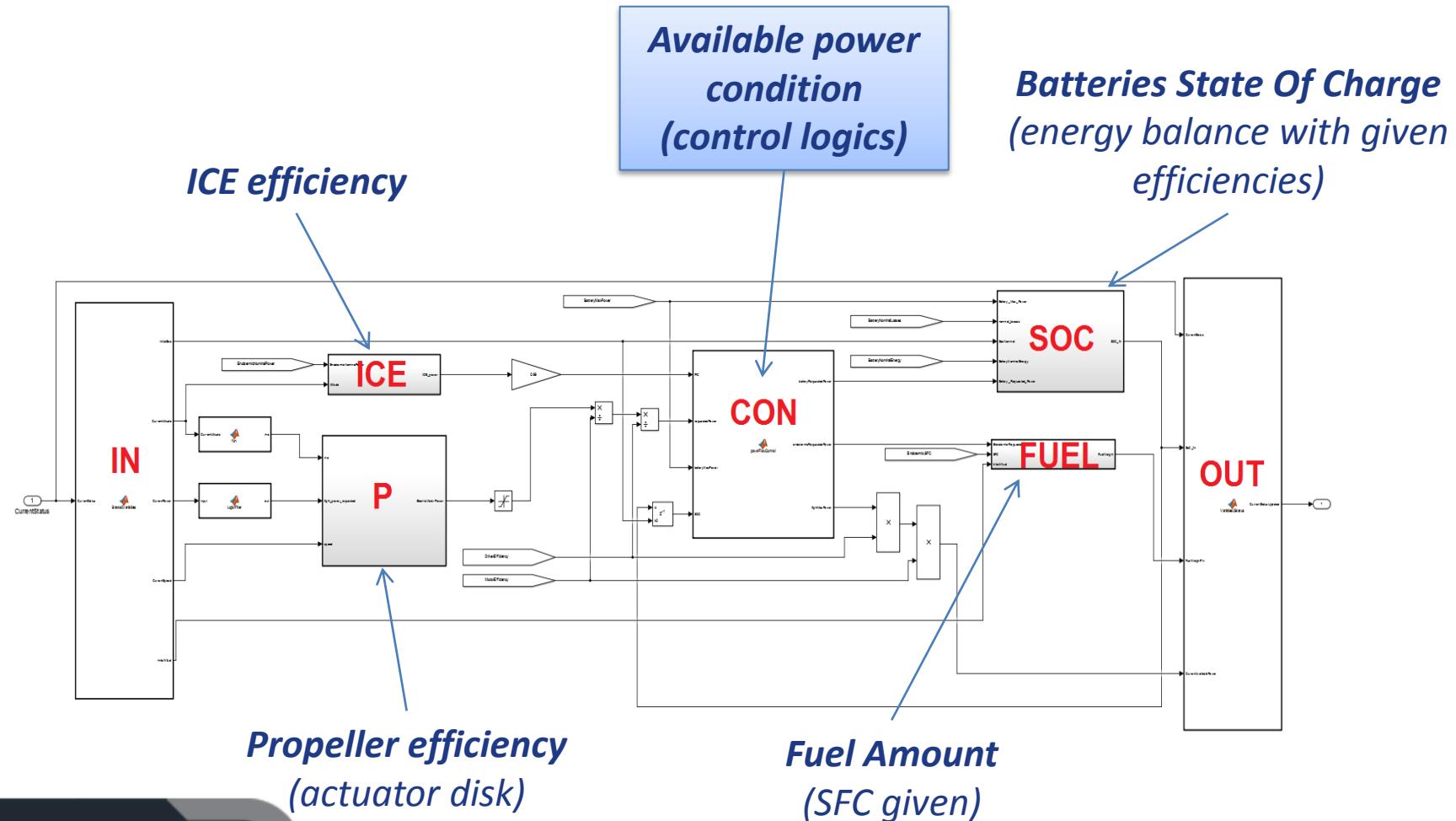
Performance Model  
(Matlab/Simulink)



Flight simulator  
(X-Plane 10)



# HyPSim: powertrain module

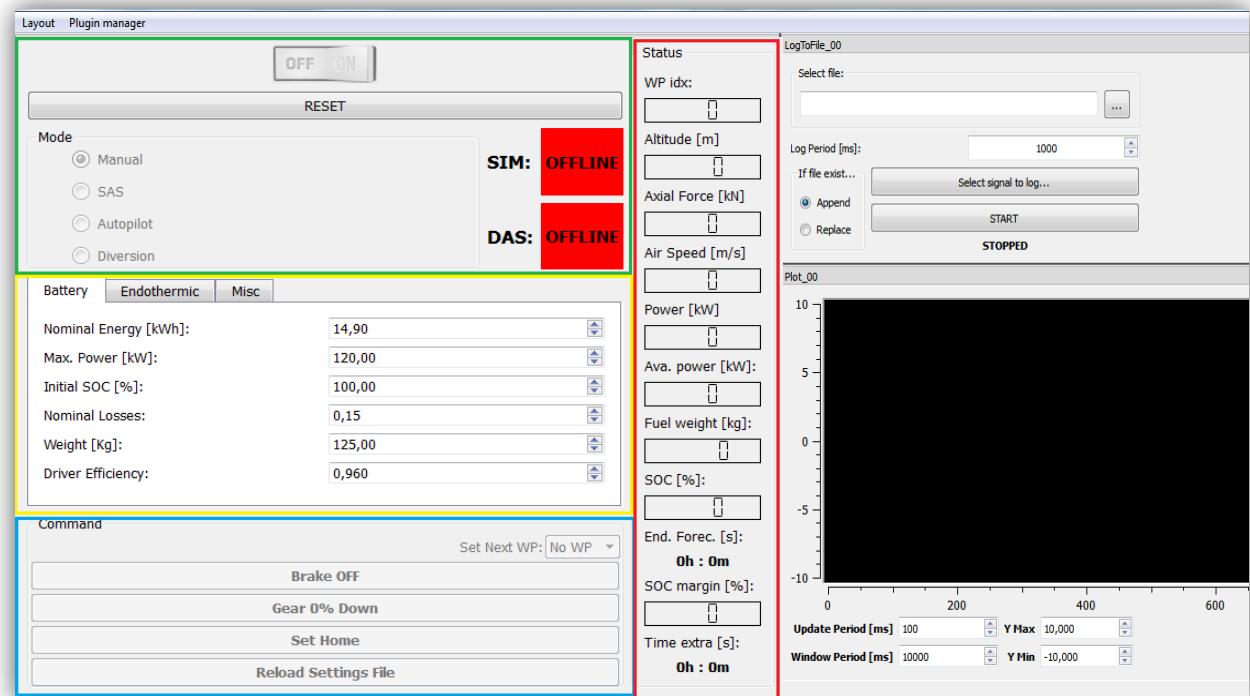


# HyPSim: the flight planner

In house software mainly developed to control the flight in automatic mode. Then, several features are added by means of plugins that can be activated/deactivated by the user.

## Plugins

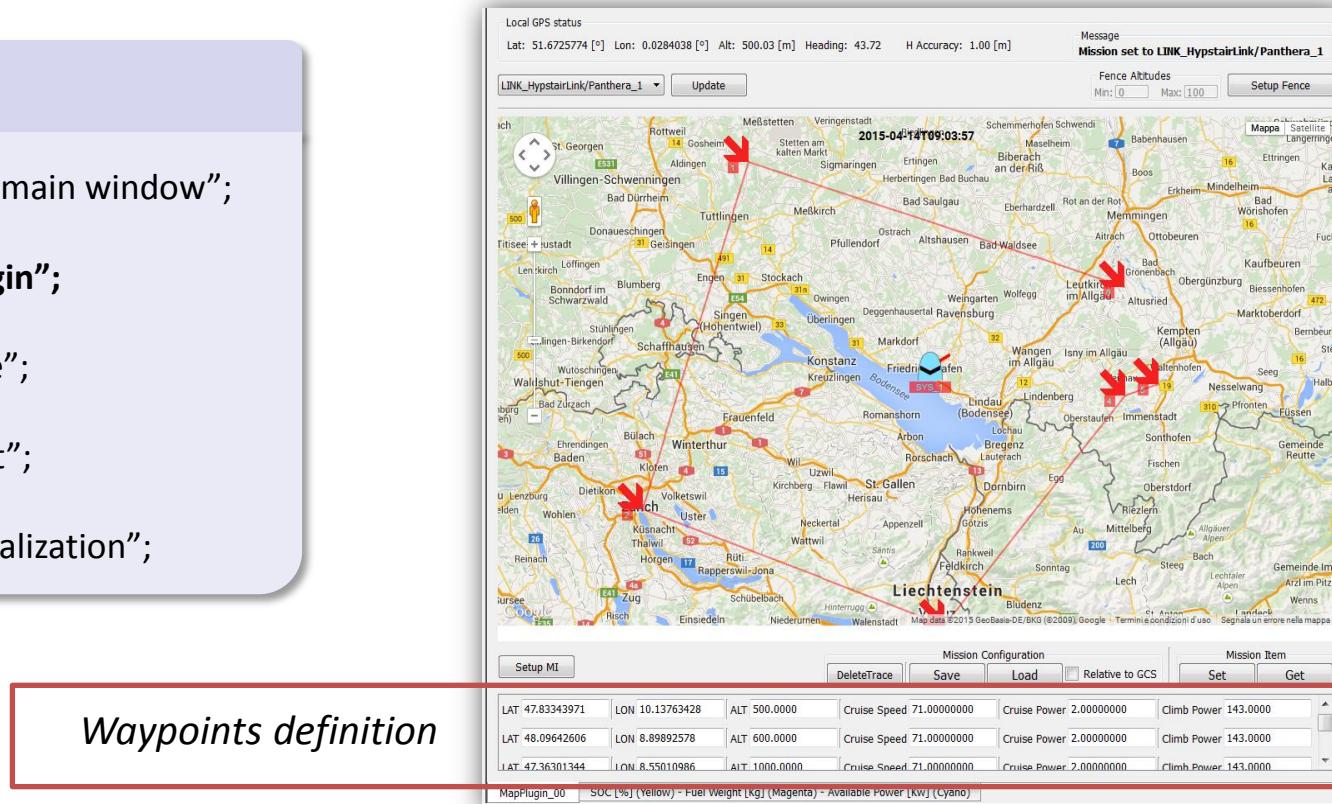
- “Hypstair main window”;
- “MapPlugin”;
- “LogtoFile”;
- “Data Plot”;
- “HMI visualization”;



# HyPSim: the flight planner

## Plugins

- “Hypstair main window”;
- “**MapPlugin**”;
- “LogFile”;
- “Data Plot”;
- “HMI visualization”;



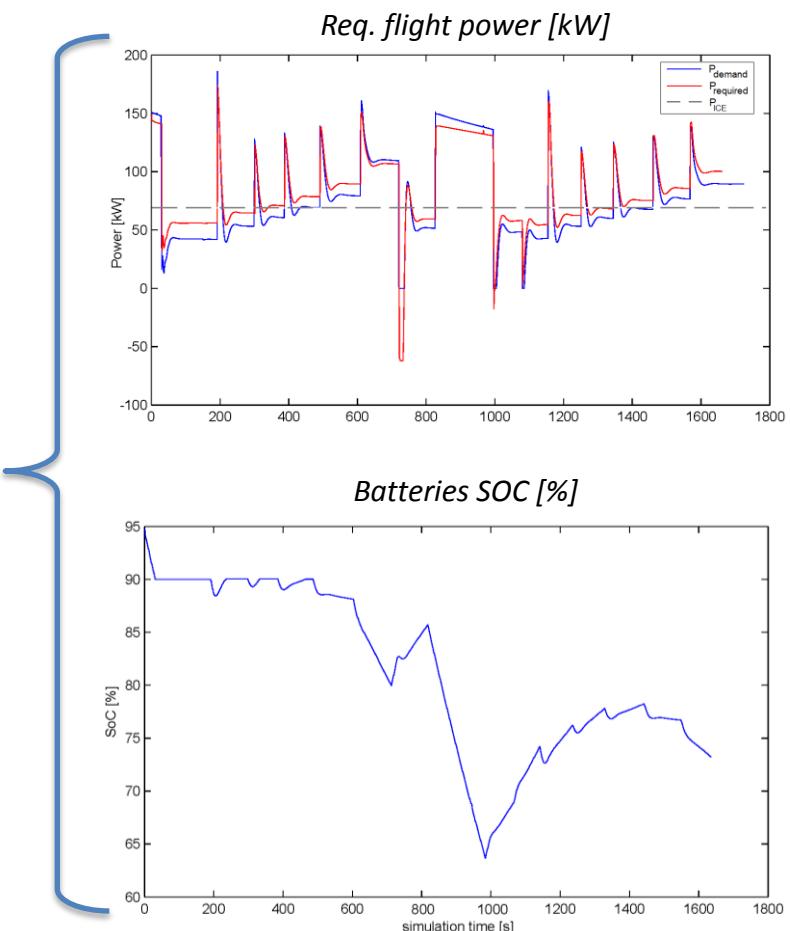
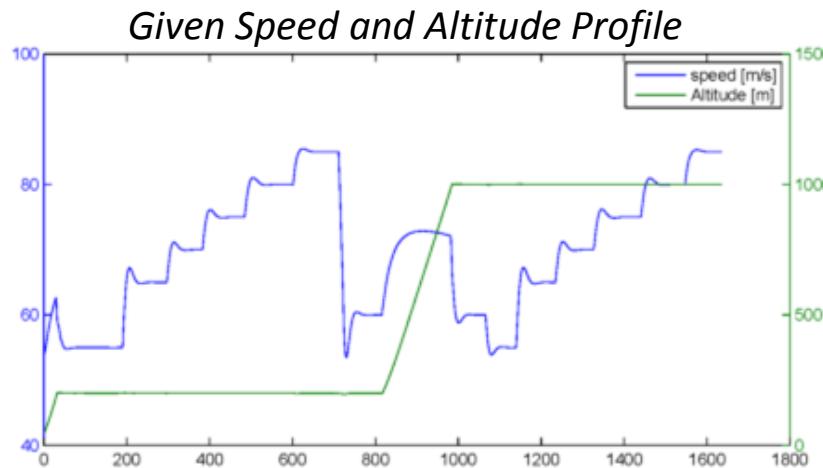
Let's have a fly!



# Example of results

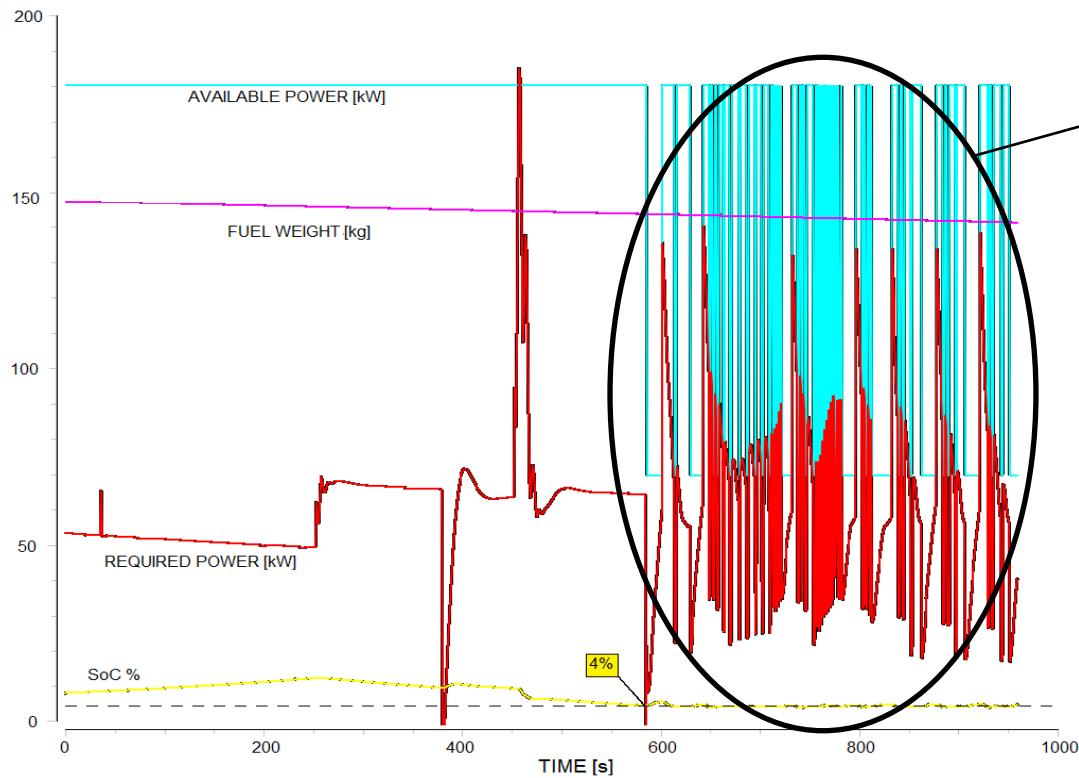
TEST CASE 1: Required power evaluation in steady and dynamic conditions (straight flight)

$$P_{req} = V \cdot D + V \cdot W \cdot \sin \gamma + V \cdot \frac{W}{g} \frac{dV}{dt}$$



# Example of results

## TEST CASE 2: Fully discharged batteries



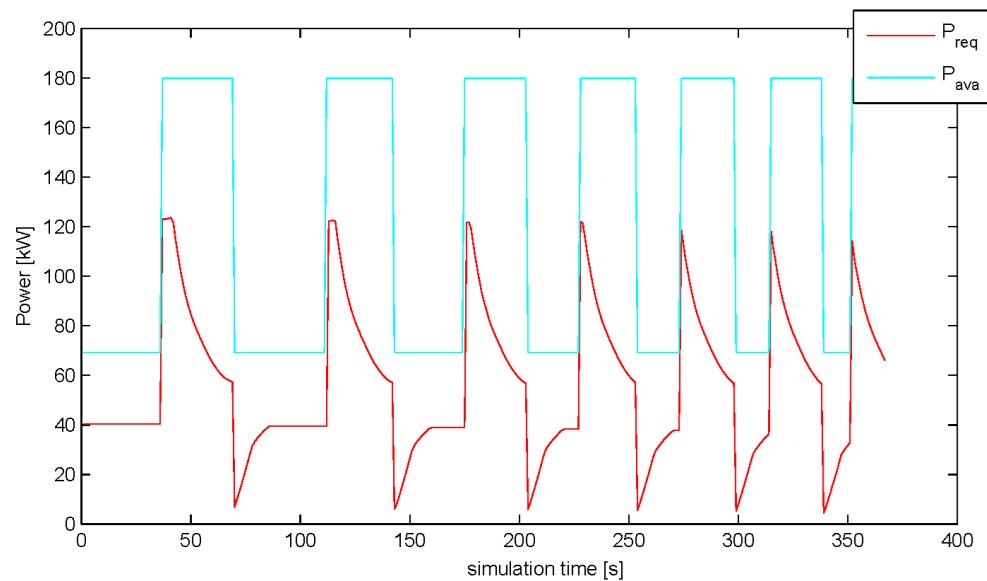
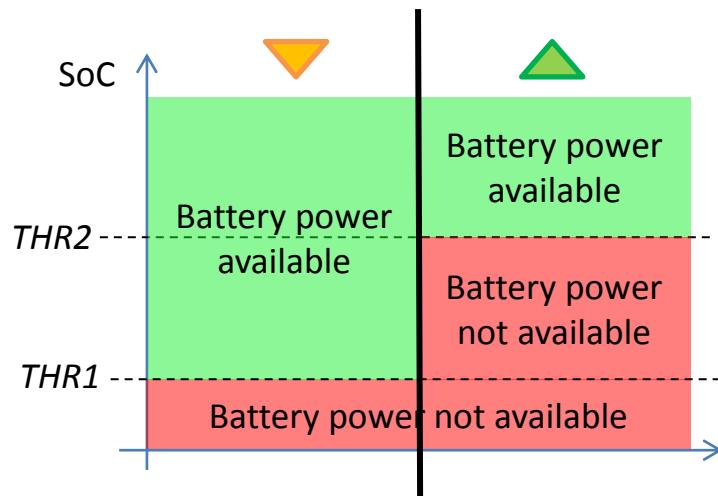
### Looping problem:

- High speed requirement
- Power required > 80 kW
- Fully discharged condition ( $\text{SoC} < 4\%$ ) is reached
- Available power is reduced to 80 kW (only ICE generator available)
- Speed is reduced, hence required power
- Batteries start recharging
- High speed requirement restored

# Example of results

## TEST CASE 2: Fully discharged batteries

Introduction of a 2nd SoC threshold (e.g. 7%) to reduce charge-discharge cycle frequency



# Conclusions

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- Flight performance analysis is essential to assess the potentiality of the hybrid and all-electric propulsion systems.
- The definition of optimal energy management remains an open problem.
- A simulator has been set up, reaching the following objectives:
  - HyPSim allows the interaction between the performance models and a human pilot
  - Instantaneous performance can be evaluated and prediction can be performed
  - Main results can be visualized through the HYPSTAIR *HMI*
  - The system has been tested for several conditions
  - HyPSim can be used as a design, training and dissemination tool.

# Conclusions

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- Further development
  - Implementation of more detailed models for the power train simulation
  - Implementation of additional control logics for off-design conditions and further testing activities
  - Deeper and more complete implementation of the HMI module
  - Integration with *haptic* input devices developed within the HYPSTAIR project

# Thanks for your kind attention

Contacts:

Dr. Fabrizio Oliviero,  
Department of Civil and Industrial Engineering, Aerospace section  
University of Pisa  
[fabrizio.oliviero@for.unipi.it](mailto:fabrizio.oliviero@for.unipi.it)



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